

AVIATION

The Oldest American Aeronautical Magazine

MARCH 30, 1925

Issued Weekly

PRICE 10 CENTS



Martin Bomber towing Aerial Target

VOLUME
XVIII

SPECIAL FEATURES

NUMBER
13

- THE AERIAL MERCURY
- N.A.A. NATIONAL AIR POLICY
- NEW AIR BALANCE AND SMALL WIND TUNNEL
- THE CALCULATION OF LONGITUDINAL STABILITY

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MARCH 30, 1928

AVIATION

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The Air Mail Division of the United States Post Office Department is the greatest exponent of commercial aviation in the world. Having successfully proved over an operating period of eight months that a day and night schedule between New York and San Francisco was feasible and practical, the next step was to obtain aeroplanes that, with the same faithful Liberty motors and with the same or better performance, would double the pay load—thus doubling the income without materially increasing the operating costs. The Curtiss Aeroplane and Motor Company, Inc. were quick to see the opportunity and developed for the Air Mail Service, the Curtiss Carrier Pigeon.

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High speed 128 MPH	Cruising speed at 1800 RPM 108 MPH
Landing speed 80 MPH	Ceiling 16,000 ft.
Climbs 1,000 ft. per minute to 3,000 ft.	

The official cross country flight of one hundred miles was made in fifty-nine and one-half minutes on twenty gallons of gasoline.

Five miles per gallon—Five cents per mile—Ten cents per ten miles.

THIS IS COMMERCIAL AVIATION WITH REAL ECONOMIC POSSIBILITIES.

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BUSINESS MANAGER

AVIATION

VOL. XVIII

MARCH 30, 1925

No. 13

Amateur versus Professional Pilots

THE safety of a professional pilot is usually no longer than it is necessary to carry a very considerable pay load and also to use the machine intensively, otherwise the cost per revenue mile of operation of a plane mounts up to a very unsatisfactory figure. The same is true in other fields where professionals have to be hired, there the cost of operation also goes up. If only professionals drive automobiles, there would not be a fraction of the number now used.

From the manufacturer's point of view the building of planes for professional pilots has a distinctly limited field if the Air Mail continued running as it present seems it would probably not need for replacement more than ten or fifteen modern planes a year. Even looking forward to a vast expansion of aerial transportation, it may be many years before the requirements for this work become as large as those of the Army and Navy. The manufacturer's field for planes piloted by professionals but used in special lines such as aerial photography, passenger carrying, cotton dusting, taxi service, fire patrol, etc. will probably enlarge in the future, but even these fields will only need a limited number of planes.

If one can judge from the automobile and motor boat industries, the largest potential field for planes in the amateur market who does not expect to earn his living out of flying but uses his plane incidentally, either for pleasure or as a convenience in his business. At the present time there are few amateur fliers that is more who have not or do not want to make money out of aviation. As a rule the high cost of flying and the lack of landing facilities are given as the reason. These are undoubtedly the main deterrents but there is another which has been considerably neglected.

The present day plane is designed for the professional and the plane that is suitable for the professional is not necessarily suitable for the amateur. The professional uses all kinds of instruments which the amateur would not be able to understand, much less use, and above all the professional wants maneuverability. He, also, wants a plane which will respond immediately to his slightest control whether he is flying level or in the most complicated maneuver. To get this maneuverability, the stability of the airplane must be sacrificed. A plane which follows the pilot's slightest whim, instead of memory staff, span, side slip, nose dive, etc. In the hands of a professional, these qualities tend toward safety. When the test pilot tries a new plane he reports on its maneuverability or unmaneuverability depending on whether or not it has these qualities.

The professional pilot does not realize that the qualities which seem safety to him may seem danger to the amateur and the dangers of the plane usually takes the professional's word. The amateur can not rely upon to within a few feet of the ground as in to make a slow landing in a small field.

Ability to stall is a danger to the amateur. A machine where a slight twist of the wrist will put it in any position is not ideal for the amateur. A certain degree of inherent stability will probably make a safer machine for the amateur than the highly maneuverable machine, which will do exactly that the amateur can not use.

It is difficult for the professional pilot to realize the limitations of the amateur pilot but the subject is of sufficient importance to warrant and study, for until a plane is built which is safe for the amateur there will only be a limited market for specialized planes.

Secretary Daniels a Little Late

THE most surprising comment in the expression of Mr. Daniels, Secretary of the Navy during the War, and for eight years under President Wilson. He writes, "The patriotic mandate in the American people is unassimilated in the air. If airplanes cannot work downstairs today, it is only a question of time when they are. We shall not far day see it 'very terrible' than at any time before."

It will be remembered that Mr. Daniels offered at the time of the bombing of the *Carrizal* warships to stand on the dock while "Admiral" Minami, as the Secretary understood him, dropped bombs from his *host* of surprise. Also, it will never be forgotten by those who were arguing some separate bureau in the Navy Department to handle seaplanes, how definitely Secretary Daniels opposed such a change. Daniels was following the advice of his advisers, just as all heads of the Navy have. It is generally thought that the most effective smoke screen that the Navy has is around the naval chief. Evidently it has been lifted in the case of Mr. Daniels and time now as we may possibly hear that he is recruited that airplanes can sink battleships.

Stability Calculations

IF this were, Mr. Kormann-Krokhovsky gives the detailed calculations for determining the degree of longitudinal stability of an airplane, following out the plan developed in his previous article. To anyone who has endeavored to make use of the formal theory of Breyne and Bussiere, this method will seem as a godsend. It has been said that an efficient engineering department could design and test an airplane before its stability could be calculated. Given the data, it is possible for anyone with a fair knowledge of mathematics, to determine the stability of a new design with a high degree of accuracy. The use of this method of calculation should considerably improve the aerodurability and stability of our light planes, as they are usually made "at home" and are of entirely new designs. We would be interested to have from any of our readers who use the method presented in this and succeeding issues.

LAWRENCE F. O'NEIL
EDITOR
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CHARLES T. ALLEN
CONTRIBUTING EDITORS

The Aerial Mercury Air Mail Plane

Description of New Plane With Many Unique Features,
Specially Designed for Mail Service

By NORMAN MEADOWCROFT, A.F.R.A.M.

The first "Aerial Mercury" has just been completed and tested by the Aerostar Service Corporation, at Hammondsport, N.Y. This company has been in business over five years.

This machine was designed by Harvey C. Mansouri, who was for many years a designer for the Curtiss Company, and is a leading designer of light airplanes, having won trophies in this class at the Empire Air Meet last year. The sky was the limit under the supervision of Joseph F. Minoli, who has had many years' experience in aircraft design, the Curtiss Company and others, both in this country and abroad.

A notable feature is that the plane was designed and built in three months, thus being quite remarkable considering the fact that it is a plane in the 6600 lb. weight class. (Daily Island) and embodies many new and desirable features.

Another feature is the provision of yards in each lower wing, which allows the pilot to see the landing wheels at all times, thus avoiding perhaps a bad crash in the event of a hard top, which the pilot would otherwise be unaware of.

Power Plant

The engine is a standard 400 hp Liberty, 12 cylinder, in a steel tube frame, having adjustable radiators, and ball joint at the junction to the levers. The complete power plant is easily removable as a unit, for quick service. A small service hatch is conveniently located in the center frame, a fine wall between it and the main compartment.

The main compartment has a capacity of 56 cu. ft., and is completely lined with sheet aluminum and fitted with leather bottoms to take the wear of the mail sacks. A large, well-hinged door provides quick and easy access to the compartment, and is fitted with improved lock and two heavy straps.

The rear compartment has a capacity of 32 cu. ft., and is completely lined with sheet aluminum and fitted with leather bottoms to take the wear of the mail sacks. A large, well-hinged door provides quick and easy access to the compartment, and is fitted with improved lock and two heavy straps.



One half view of the Aerial Mercury control surface and exhaust baffle. H. C. Mansouri is in the cockpit.

pair of ailerons. The wing beams are of built-up box sections, the sides being formed of two-ply diagonal spruce planking, all glued and nailed together with galvanized nails. The ribs are of laminated type, and the completed wings have been cut and balanced twice, so that no lateral pressure to render them inactive need be applied to render them inactive proof. On test, in the short time of assembly, the aileron, one rib having an actual weight of 32 oz., held a distributed load of 10000 lb. before breaking, thus demonstrating the large factor of safety of its construction.

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The pilot's cockpit is roomy and comfortable, being lined with veneer, and having an adjustable seat. Compartments are provided for carrying log books, maps, and landing flares, and a cupboard behind the seat allows for the storage of luggage and tools. Instruments are fitted in suitable positions on the instrument board, all being marked in stainless steel. The control is by means of a four-spoke stick, which in this machine is of dual-axis type, and has a convenient switch in the end, for illuminating the board as may be required.

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The aeroile starting the pilot, as would permanent lights mounted on the board in the usual way. These emergency controls are fitted to the board, one for releasing all gaslines from the main tank, one for releasing a propeller blade in the tail and side, for use in case of a ruptured landing field and another to release pyrene fluid from a tank at 100 lb. pressure which is opened over the engine from pipes and stored off the main tank.

The control wheel operating the rudder surface adjustment is also located conveniently to the pilot on the right side of the cockpit. All engine controls and wiring are run along the fuselage in a metal tunnel, making a neat and compact arrangement. A handle taken from the earliest types, carries the propeller, and the propeller is held in position by a wire which passes through the pilot's cockpit, and giving heat controlled by a valve placed within easy reach. This is a desirable feature for flying in severe weather.

An aluminum oil tank of 10 gal. capacity is mounted with an air cooled aluminum oil-cooler. The radiator is of corrugated aluminum, fitted with a thermostatic valve, and mounted centrally on rails above the engine. All filter cups are of the quickly removable, snap type. The main gasoline tank is strong enough the fuselage under the main compartment is shaped to follow the lines of the fuselage. It is fitted with a 1-in. diameter valve, but can also be arranged for quick release when desired. The tank holds 100 lb. of gas, and the upper wing at a 14 gal. gasoline tank, securely mounted, and easy of access. The overflow from the gravity tank is carried to the trailing edge of the wing, where it is in full sight of the pilot.

For use when it is desired to carry a passenger, a portable seat of wood frame, front opening, is provided by folding and unfolding and having a comfortable canvas seat, is supplied. This fits into brackets secured on the levers in the main compartment, and a removable windshield is also supplied in conjunction with it.

Shock-absorbing wing tip-shocks are fitted, of cast design, and are secured to the body of the fuselage. Proper absorbing always a difficult arrangement to work out satisfactorily. In this case, the object of care detracted, with absolute safety and freedom from breaking away in flight, has been achieved. The landing gear is very robustly designed, depending for its shock-absorbing qualities on the rubber shock absorbers. The tail and shock absorbers are taken in the same manner, and the dangers of deterioration and breakage as appeared in the much-revered rubber cord shock absorbers, are eliminated.

The plane, as built and described above, has a high speed of 135 m.p.h., and a landing speed of only 47 m.p.h. If the two sets of smaller lower wings are removed, and a single pair of smaller lower wings is supplied, or arranged as to use the main wings and fittings as the right plane, and on-pitch of wing changed in about half an hour. This plane then has a high speed of 130 m.p.h., and a landing speed of 54 m.p.h.

After four short preliminary trial flights at Hammondsport, the eight plane was flown down from Hammondsport to

Hancock Field on Friday, March 6, making the journey of 300 mi. in exactly 2 hr. in very notable performance. On this trip the pilot carried pilot H. C. Mansouri, Joseph F. Minoli, as passenger, with their personal baggage, and started with a full tank of fuel.



Top—Four men standing outside aircraft. Bottom—Five men about

Characteristics

Weight	Gross weight	Empty weight	Passenger weight
Weight	4240 lb.	3400 lb.	840 lb.
Empty weight	2400 lb.	1775 lb.	625 lb.
Gross weight	4240 lb.	3400 lb.	840 lb.
Passenger weight	-	-	-
Wing area	300 sq. ft.	300 sq. ft.	300 sq. ft.
Span	35 ft.	35 ft.	35 ft.
Wing span	35 ft.	35 ft.	35 ft.
Wing aspect ratio	4.5	4.5	4.5
Wing loading	11.5 lb./sq. ft.	11.5 lb./sq. ft.	11.5 lb./sq. ft.
Wing thickness	1.5 in.	1.5 in.	1.5 in.
Wing chord	4.5 in.	4.5 in.	4.5 in.
Wing camber	0.5 in.	0.5 in.	0.5 in.
Wing twist	0.5 in.	0.5 in.	0.5 in.
Wing incidence	0.5 in.	0.5 in.	0.5 in.
Wing dihedral	0.5 in.	0.5 in.	0.5 in.
Total vertical lift	-	-	-
Downlift	-	-	-
Drag	-	-	-
Wing drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference drag	-	-	-
Wing induced drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference drag	-	-	-
Wing induced drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference drag	-	-	-
Wing induced drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference drag	-	-	-
Wing induced drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference drag	-	-	-
Wing induced drag	-	-	-
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Wing induced drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference drag	-	-	-
Wing induced drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference drag	-	-	-
Wing induced drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference drag	-	-	-
Wing induced drag	-	-	-
Wing parasite drag	-	-	-
Wing profile drag	-	-	-
Wing wave drag	-	-	-
Wing interference			

Certified Financial Statement

The following is the report of the president of the Custer Aeroplane and Motor Co. to the stockholders and dated March 2. It shows an improvement over previous years and is noteworthy and contains a statement of new designs and products under way.

To the Stockholders:-
The financial results of the operations of your company for the year 1958 are summarized herewith. These results show steady progress as the liquidation of capital employed in the business, along the lines laid down in the annual report for the year 1957.

REPORT THE PAST YEAR
The past year the company has reduced its funded and other indebtedness \$36,617,334, while the reduction of cash assets other than securities, during the same period, has been only \$13,676,762. The working capital position of the company is stronger than it has been at any time, there being practically no current liabilities. The Cuddeback Field at Macomb and the experimental plant at Goodrich City have been closed, and no foreign debt.

The net earnings after full depreciation amounted to \$10,625.18, a large part of which was derived from commercial operations as distinguished from Government contracts. The Directors have declared, for the period, dividends of \$4.90 on the preferred stock of the company.

The sharp competition from Government plants, and the destructive competitive bidding on Government contracts, resulted directly in the loss of many Government contracts during the first part of 1944, especially from the Navy. Toward the end of the year there was a distinct betterment in both respects, especially in respect to Army contracts, due to the fact that the lessons of the Army Air Services made an intelligent study of the matter and, in view of their own interests, recommended to make a honest study of warfares as it exists at present. The work of this committee seems to be distinctly remunerative and its predecessor's statements are so indicate that Congress will take a stand against the destructive tendencies that have grown up in the War Services since the close of the War.

In the last year's report, your company has succeeded on model basis in engineering advances as it had anticipated prior to last fall; nevertheless, despite the year, you graduated some important advances in the art. The development of the steel metal propeller has been pushed forward steadily and is now a substantial source of revenue to your company. Two standard commercial types of airplanes have come out—the Douglas D-24 and the Cessna 140, which is a major step for the Post Office Department and the other air mail delivery services. Both show fair promise of success. In the military types the Curtiss CG was adopted during the year by the Army or a standard light bomber, scout and torpedo plane, and no production contracts resulted to your company owing to the fact that your company was unwilling to build these airplanes before what it considered a fine price was obtained. Your company has, however, developed one of the best engineering models that your company has produced during our history and promises to be a good source of income, both at home and abroad, for some time to come.

In the development of motors, scientific work has been done by the engineering division of your company. For the last three years your company entered the field of the air cooled motor and purchased, for the Army, a type of large power. The motor is still early in the development stage but enough has been learned to indicate that it has large possibilities. The first motor was the 10-12. One hundred were delivered during 1924 and twenty-five the fifty horse issue just at February, 1925. This motor, while some fifteen pounds lighter than the 12-12, delivers approximately 25 per cent more power and is now advised by recognized engineers in its class as good as anyone it was the 12-12 itself, which brought back to the United States the speed records of the World and held them

It is not possible at this time to outline very definitely the policies of your company for the coming year. Aviation is in a completely uncertain state. After five years Aviation

CURTISS AEROPLANE & MOTOR COMPANY, Inc.,
C. M. CURTIS, President.

What One Newspaper Is Doing for Aviation



The red appears at the head of a weekly column of news in the Boston Evening Transcript. This effort deserves to be magnified.

Stability and Controllability of Airplanes

Appendix I—Method of Calculation

Dr. B. V. KOROVIN-KROUKOVSKY

From the earliest days of aeronautics and up to the present stability remained the most troublesome question for all experiments or present aeronautical engineers. Despite large amount of literature on the subject and great efforts of science there is still no simple method available by which the designer can estimate the stability of the surface plane stability. Both wind tunnel experiments and the airfoil theory show that for the moderate angles of attack the moment, lift and center of pressure coefficients are connected together by the simple relation

where C_0 is the constant coefficient defined as the moment about fuselage axis divided by the quantity $\frac{1}{2} \rho V^2$.
 C_1 is the lift coefficient defined as $1/2$ divided by ρV^2 .
 ρ is the dynamic pressure = $\frac{1}{2} \rho V^2$.
 ρ is the density of air expressed in pounds per cubic foot.
 s is the wing area in square feet.
 V is the velocity of flight in feet per second.
 r is the distance of center of pressure from leading

stability characteristics is very doubtful, except when the noseplane repositions the slight trim moment resulting from the angle of attack. It is also found that the center of gravity is far back as 25 per cent of the chord, others give breakups as 20 per cent of the chord, others again give breakups as 20 per cent. It is realized that the center of gravity is located higher with respect to the tail at most also be removed, but the exact relation is unknown. The effectiveness of the tail is still about 50 per cent and although the importance of the aspect ratio of the tail is not known, it is believed that a decrease in aspect ratio will increase the stability. The effect of the position of the center of gravity with respect to the tail is not known.

cept in the propulsive thrust line. Then we see that although the importance of different characteristics on the airfoil is realized and consequently discussed quite fully, no information as to the effect of the angle of attack on the airfoil is given. In the present work we will attempt to find the value of such information, and to derive the method by which the degree of stability of an airfoil can be predicted from its design characteristics, and will demonstrate the actual use of the method by applying it to existing airfoils. Of course we will have to limit ourselves to the consideration of only the most important characteristics of airfoils and to resort to the use of the first few polynomial approximations in order easily to supply the necessary accuracy. Yet we will be able to show that calculated results agree well with the properties of the machines as reported on the flying tests.

Airflow Feasibility

The aerofiles of the type used for the supporting surfaces known to be unstable, i.e. if balanced about some point on the chord while being subjected to a stream of moving air and left free to rotate they will tame over. To be able to discuss the aerofoil as such the angle of attack which we measure is the point of rotation of the chord about which the aerofoil rotates. This means that the point of rotation of the chord would be from 35 per cent to 50 per cent of the chord from the leading edge, and this is where the center of gravity was usually placed on the old time airfoils. These results, however, that the center of pressure is only slightly unstable, namely the point of intersection of the chord and the center of pressure moves in the direction of the air flow, though the angle of attack is at very small angles that such a motion occurs, as probably more stable airfoils were produced by the attempt of the designers to locate the center of gravity at some position of the center of pressure than by any other cause. We will see that the position of the center of pressure is not the same as that of the center of gravity.

The importance of an accurate knowledge of the quantity was, no doubt, recognized by L. Pfeiffer and A. Döderlein [Die Technische Mechanik, Berlin, 1911], who give it the value of 0.01. More recently (see, for example, Ritter, 1925) and the value it is as 0.01, the value which we have used to check very steadily for the acoustic popular among American engineers.

for the tail-surface, which have neither lift nor moment about leading edge at zero angle of attack, and have practically stationary center of pressure at 20 per cent of the chord.

We are from the above that for stability of a tailless airplane its center of gravity would have to be placed not farther forward than 20 per cent of the chord. This is true if we assume standard surfaces giving good lifting capacity more than 20 per cent or 30 per cent of the chord such as surface α which he evidently had to put out of balance. In order to balance it we will have to add tail surfaces, subjected to down pressure of the air. Now, the cambering besides balancing also gives a little stability, which would allow us to move the center of gravity only a little further back, namely to 28 per cent of the chord. This is done by the cambering effect of the tail, and by the vertical division of the center of gravity from the wing and thrust line, which we are prepared to do now.

The sketch given here shows the forces and moments acting on an airplane in flight. The force of gravity acts of course through the center of gravity and is not shown, as producing no moment it is of no interest to us. The applies to the components of gravity perpendicular to the line of flight and parallel to it, the latter occurring during steady flight. We will consider the effect of the moments of the machine of rigid bodies we will consider all moments and all rotations taking place around the center of gravity. The effect of the head resistance on stability will be neglected entirely, or in other words it will be assumed that it acts near to the center of gravity and varies nearly as dynamic pressure C_d . The thrust line is assumed to be parallel to the line of flight, which is often a good approximation at sufficient angles of attack. We will now proceed to find the effect of the tail on either a simple and practical working method we must look carefully in the consideration of only the most important issues and effects. In accordance with this statement we will consider only the moments generated by the change of the angle of attack and will neglect the moments due to vertical and horizontal translational velocities resulting from pitching. The effect of these velocities can be shown to be small compared with that of the angle of attack, and can be considered another of Breyer and Farncomb. We will, however, make exercises in the analysis of the effect of propeller thrust, where the change in velocity is the most important factor.

Detailed Calculation

The wing shown on the sketch represents the wing of a monoplane of constant chord, or mean chord of a monoplane with the tapered wings of a biplane. In other words it is the wing at an imaginary monoplane equivalent in all its properties to the actual wing combination of the airplane under consideration. For the convenience of reference we will introduce new the notations used in this work and not explained earlier.

- C_l — Lift coefficient of the wings.
- C_m — Moment coefficient of the empennage.
- M — Weight of the airplane in pounds.
- P — Horsepower of the engine.
- S — Wing area in square feet.
- s — Cambering (camber and deflection) area in square feet.
- b — Overall span in feet.
- b' — Empennage span in feet.
- G — Mass gap in feet.
- α — Angle of attack of wings in degrees.
- β — Angle between the chords of wings and empennage.
- γ — Angle of attack of the empennage.
- δ — Angle of deflection of the empennage.
- N — Effective aspect ratio of wings = b^2/S .
- N' — Effective aspect ratio of the empennage = b'^2/S .
- F — Lift of the wings = $q S C_l$.
- f — Lift of the empennage = $q S' C_m$.
- M_{sum} — Sum of the moments of all forces about C.G.
- $M_1, M_2, M_3, M_4, X, Y, Z$ — the coefficients of moments of which will be explained later.

We will return the conventional notation by assigning signs sign to all driving moments and plus sign to all stabilizing ones, and we shall assign positive sign to all quantities measured as shown on the sketch, i.e., the center of gravity located below the wing and the thrust line, and the vertical stabilizer located below the empennage line from the wing chord. This settled we can write now the moment equation for the diagrammatic airplane as follows:

$$M_{\text{sum}} = (x - p) F + fL - fL - Tg$$



Let us assume now that the airplane is pitched through the small angle $\Delta\alpha$, corresponding to a small increment of the total vertical motion $\Delta\alpha$. Several of the quantities in the above equation will undergo small changes, and neglecting the small quantities of the second order we can write a new moment equation for the diagrammatic airplane as follows:

$$\begin{aligned} M_{\text{sum}} &= (x - p) F + \Delta(x - p) F + (x - p) F - fL \\ &\quad - fL - Tg - \Delta Tg \end{aligned}$$

In writing this equation we assumed that the center of gravity on the upper surface does not move, the assumption being justified for the symmetrical sections. Redrawing the first equation from the second we get:

$$[\Delta] \Delta M_{\text{sum}} = (x - p) \Delta F + \Delta Tg + \Delta(x - p) F - \Delta fL$$

We will call the second coefficient of the derivatives the quantity $M_1 = M'_1 q S C_l$, and will endeavor to examine and expand each of the increments in the above equation in time, in such a way as to be able to write down the derivative of M_1 with respect to the lift coefficient C_l , which will tell us the stability coefficient and will dictate $-N$. In doing this we will find the quantity q as constant.

The lift of the wings can be written:

$$\begin{aligned} F &= q S C_l \\ [\Delta] \text{ hence } \Delta F &= q S \Delta C_l \end{aligned}$$

In order to expand ΔC_l we must call attention to the fact that x and y are the ordinates of the leading edge of the wing with respect to the center of gravity, and that x is always measured along the line of flight and y perpendicular to it. Hence the increment of x will depend on the value of α and the increment of the angle of attack $\Delta\alpha$. It is easy to see that

$$\Delta x = \frac{y}{b} \Delta\alpha$$

where $\Delta\alpha$ is measured in degrees. It is known from the laboratory tests for that the aerodynamic angle of attack used in ordinary flight the lift coefficient of each section in the linear function of the angle of attack, i.e., $\Delta C_l = \alpha C_l$. Consider the value of this essential section but take for definiteness dependence already stated on the aspect ratio, so this will be expressed on detail later. In connection with the study of the tail effect ΔF presents itself well sufficiently accurate to adopt the same value of $dC_l/d\alpha = 0.5$, corresponding to an effective aspect ratio of 5. Substituting this in the above formula [Δ] we get:

$$\begin{aligned} \Delta x &= \frac{y}{b} \Delta\alpha \\ \frac{\Delta x}{180} &= \frac{y}{b} \frac{\Delta\alpha}{45} = \frac{y}{b} \frac{1}{360} \frac{\Delta\alpha}{0.5} \Delta F \\ [\Delta] \text{ or finally } \Delta x &= 0.025 y \frac{\Delta\alpha}{b} \end{aligned}$$

(To be continued)

Curtiss Wins C. O. Award

There has been a general understanding concerning the results of the Corps Observation type airplane competition that was held at McCook Field during the last month.

Early last spring the Air Service invited competitors to the development of a new Corps Observation type airplane to be powered with the Pratt and Whitney 250-hp. engine, or other suitable engines of that type. This competition was intended to be held in November and a Board of Officers was appointed to receive a report thereon. This competition was won by the Curtiss Company.

It should be mentioned that the Air Service required a plane which would replace the DH-4 powered with Liberty motors, the Curtiss and Douglas Companies were invited to submit their plans for construction along with the McCook Field type and one previously presented by another competitor, the Allis-Chalmers Motor Company. The Air Service required a plane having performance or military adaptability were not considered of primary importance. The Board recommended and the Chief of Air Service approved the acceptance of the proposal of the Curtiss Company for a new observation plane.

The Board recommended that the Air Service award the contract for the production of 100 observation planes to the Curtiss Company. The Air Service accepted the proposal of the Curtiss Company for a new type of observation plane. We quote from the official statement of the Air Service as follows:

"The War Department has awarded \$70,000 to the Curtiss Co. and \$60,000 to the Douglas Co., for the experimental observation airplane manufactured by each of these companies, and the development work involved in the production of the new plane."

"The Army Air Service is seeking a safer and more maneuverable type of airplane for general observation use but will replace the war-horse DH-4."

"The awards were made as a result of tests conducted at McCook Field, Dayton, Ohio, Jan. 25, by a board of officers selected for that purpose. Members of the board were: Col. Charles E. Lindbergh, Capt. Lloyd L. Harvey, Capt. Harry R. Bowes, 1st Lt. Carl G. Coffey, Capt. Frank B. Burns. They are all expert aviators or observers and made strict tests of the new designs of observation airplanes which have been submitted by the Curtiss, Douglas, and Wright Companies.

"The speed of the tested types was over 120 m.p.h., with wing loading over 200 lbs. ft."

"In order of grading was accepted, which included performance flying qualities, maneuverability and maneuverability features. A total of 30 points was granted for performance based on speed, service ceiling, and crossing radius, 25 points were granted for flying qualities which included stability, maneuverability, and controllability. Landing and taking off, 20 points were granted for the landing gear which included both the engine and undercarriage. 16 points were granted for maneuverability features, which included cockpit arrangement, utility of crew, cooling system, self-starter, fast-explosive, control mechanism and visibility adjustment."

"The Curtiss Company accepted the board of rating by those presenting a total of 100 points, and the Douglas came up to the requirements with a score of 100."

"The board was favorably impressed with the design that has been made in commercial airplane manufacturing and design within the past few years."

Government Competition with Commercial Companies

In the past, whenever there was a new feature that enhanced the use of airplanes to expedite the delivery of pictures or to the U.S. Signal Corps, Army, Navy and Air Mail planes have long available to the commercial airplane services for this purpose. This competition has given the idea of many minds in the companies who would have otherwise been paid for this work.

Essentially, as the result of protests by commercial aircraft engineers, the use of government airplanes for this purpose has been discontinued.

A letter from Casey Jones, manager of the Curtiss Exhibitors Company, confirms this new policy. He sends a picture of Harry Hardy, actress, from "Pattie Page," taken before a record flight made on the day of the total eclipse. "This picture illustrates how," Mr. Jones writes, "we have been compelled to compete with commercial companies in supplying aero-ports and airplane services for the country's commercial progress."

"This position has done a great deal to develop a demand for commercial airplanes and now that the commercial companies are in a position to supply a complete and quick service for the transportation of government officials, including the Air Mail, Army, Navy and Marine. However, the change of policy, with particular reference to the transportation of persons and film of the interpretation, have definitely stated that they will no longer compete with commercial companies in aerials of the nation."

"It is interesting to note the immediate results of the adoption of this policy. Within a month after the new planes left Washington for various cities, including Philadelphia, New York, Boston, Columbus and Chicago, carrying film and pictures for at least ten different newspapers and film companies. With one exception, all of these planes got through an schedule of flights in the manner that failed to get through made by the commercial field from Washington to Columbus in the face of very unfavorable weather conditions and was only prevented from continuing its flight to Chicago by rain and fog, which made sight flying extremely hazardous."

"We are enclosing a copy of a letter from Mr. Cohen, editor of "Pattie Page," who has paid himself off by getting his pictures published in the illustrated by the enclosed photographs and apparently he will give up the use of the services furnished by the commercial companies on March first."

"We are confident that the adopting of this new policy will be a great aid to commercial aviation and will encourage various companies to organize permanently with the view to handling this type of commercial work."



No more means taken from you rigs of passenger planes

My dear Mr. James:

Compliments on the splendid work of your Company at Washington. You are doing great work in transporting our messages faster. Not only can you do this but you have general knowledge of route selection, but also from personal experience in the trust concept of that little racer of yours.

Transporting those flies from Washington to New York in the record-breaking time of one hour and twenty-eight minutes is a record worth being proud of. It is a great step toward the development of airmail and the improvement of commercial aviation and I am quite certain that your achievements as these will encourage the wider use of airplanes for quick transportation.

I social especially and with much pleasure the careful preparation that you made to compete speed and safety, as all of the planes that we used. It is in this though serves that keeps efficient organization and assures success.

Very truly yours,

(Signed) E. Cohen

New Air Balance and Small Wind Tunnel

By ALBERT MERRILL,
California Institute of Technology

Before describing this balance I would like to tell a little history about my research with aerodynamic problems.

The first two types of balances in common use today are the N.P.L. and the "Hanging Wire" balance. The N.P.L. balance is very old and was used first by S. P. Langley of the Smithsonian Institute in the early months of the last century. Langley called this balance his Residual Pressure Recorder. He placed the lateral area (open) horizontal and measured the rate of ascent and descent of the resultant pressure. I believe Orville Wright in his private experiments used one who was later to become famous. The N.P.L. people simply took Langley's Residual Pressure Recorder, rotated it 90° deg so that the span is vertical, mounted it on a pivot (Langley had a gimbal mounting) and measured the Lift and Drag components. The N.P.L. balance is, of course, an improvement over Langley's.

Early Aerodynamic Research

I happen to know about Langley's work because near the close of the last century I did some aerodynamic research work for him with a grant of money from the Smithsonian Institute. We were partners on those days and my particular job was to find the center of pressure travel on a curved surface in a natural way. Of course, I got no results, as it turned out, but the amount was far beyond my usual quantitative accuracy.

About this time I was gliding in the Chasman gliders, and made trips of from 100' to 150' H. In those days we had no record surfaces. Casted in cast and pitch was obtained by shaping the tips toward the high lift.

Along with the outdoor work I built the first hanging wire balance which was used for fundamental work in 1910 as I know. I tried the balance in the old M.I.T. building on Trinity Place, Boston, Mass. Any old M.I.T. graduate who reads this will remember that in the basement of the old building on Trinity Place, apparently under the office of Prof. Harvard, there used to be a slot near the ceiling. This slot was part of a ventilation system. When I was not using the balance I would stretch out in the slot where I had a beautiful chance to get all mixed up with the mess of pipes and machinery which occupied the space. That was my wood board. Prof. Charles Lewis was a good friend of mine and was very much interested in this work and helped me. I had two boards for weighing the lift (downward) at the front and rear edge of the tail surface and a wire to the door of the slot. The balance is shown drawn around in Messing's book "Simple Aerodynamics and How Airplanes Fly." 2nd. Ed. edition, put out by one government from McCook Field in 1924. Of course I got no worth while results with such a variable F.

Some Personal Experiences

Shortly after this, that is in 1909, the Wrights began their flights. I became well acquainted with their work through Octave Chanute. When I founded the Boston Aeronautical Society, in 1909, Mr. Chanute was its first president, and he was a very valued friend of mine from that date until his death. I had some correspondence with Wilbur Wright about their work and two days after they made their first power flight they wrote me a long letter完全是 personal and said they wanted me to help them. Immediately I wrote as outlined on the flight which was published in the Boston Herald. I felt then that the brothers were much more capable of handling the subject than I was, so for several years I did not work in aerodynamics, being satisfied simply to watch developments. In 1910 I started a series of lateral surface tests on the tail surfaces of Wright's biplane, and which I sold to the Wright Co. in 1911. In 1912 I started some experiments which I had to stop owing to lack of funds and I am only just now completing these experiments.



Mr. Merrill standing behind the new air balance and wind tunnel which he developed at C.I.T.

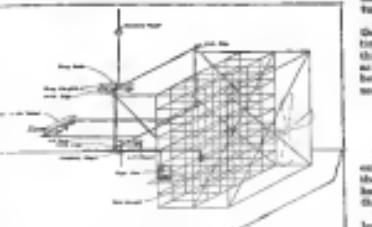
Characteristics of New Balance

Requirements 1, 2, 3, 4 are satisfied with the balance I am going to describe and be a slight improvement I expect to satisfy No. 5 too. I will now describe the balance. Fig. 1

March 20, 1935

AVIATION

is a diagrammatic view of the balance. Fig. 2 is a photograph of it. It is designed for model testing. I believe we have done quite enough work on surfaces. The M.I.T., N.P.L., and Göttingen Laboratories have turned out very good tools of



Diagrammatic sketch of the Merrill air balance and wind tunnel

all kinds of surfaces, but very little model work has been done and the balance is very simple, a clean balance and has been the answer for drag research. Take the laboratory focus on parts and sum them up for the whole model. This has produced errors, because each a method entirely neglects the mutual interference of the parts which in some case may be very great.

In my balance the model is made with its own e.g. propeller blade and also perfectly. The pitch radius of gyration must be measured with a dial gauge as the latter can be figured. The model is mounted, which is span horizontal at an outer point parallel to the lateral axis, so a frame which passes forward to a cross slot. This frame has a zero weight to balance the model weight and a weight for compensating lift. This system rests on another frame, which is span horizontal at a point 1/3 of the span from the trailing edge. This latter frame also has a tail section which allows the test frame with the model to be displaced downward and a weight measures the drag. Because of an increasing the model has to rotate until the vector passes through the lateral axis. Photographing the model against a frame of reference gives the angle of incidence. The speed of air is measured with some type of anemometer.

Operation of the Air Balance

To operate the model is set with a known setting of the elevator, the air is turned on, the weights are displaced until D and E are offset and the operator then takes a photograph of the experimenter when he has a record of L.D.V., angle of incidence and point of application of the vector, (e.g.). This gives the static characteristics. When the pitch dynamics variables are made an arbitrary guess. When the gain makes the model start oscillating we offset the pitch again to get the oscillations die out. By means of a moving picture we should be able to get the magnitudes of the original oscillations and the damping time. Of course, we have to correct for the drag and the external portion of the balance.

What I like about this balance is its low first cost, and I believe very well. My own balance is estimated to cost \$1,000.00. A four foot span model would be estimated in a cross section of from 2.2 to 8.6 for different angles. By closing the air speed I have made a paper model take off, climb rapidly, level, level and land properly. I have also suspended the tail section with the tail machine is the same type and will have something to say about this earlier later.

The size of my fan is only 12 in. and my speed is only about 2 m./sec. The variation over the testing surface how-

ever is less than ± 2 per cent in spite of the tests being on the delivery side. The models I have been using are of 8 in. span. I think that even this small model will give good results in qualitative work with a series of models. When the best of the series is found it can be tested in a large wind tunnel.

I hope that the younger generation of enthusiasts will build this kind of wind tunnel and do research work. Truth is the time of investigating and inventing and we need both of these things right now in this game. It is an unfortunate fact that as man grows older the period of youth has hence tends to become less and this is next to impossible to make less use value in a new idea.

Membership in R. Aero Society

In view of doubts which appear to exist as to the qualifications required from candidates for Associate Fellowship of the Royal Aeronautical Society of Great Britain, the Council has issued the following statement as to the interpretation of the Regulations.

The Regulations provide for the election to Associate Fellowship:

- (a) Those who pass the Society's own examination, or possess an equivalent University or Technical College qualification, followed by 2 years' practical experience in the application of the science of aeronautics.
- (b) Those who pass a Thesis.
- (c) Those who have no exceptional qualifications or achievements but do not possess the examination qualifications outlined in (a).

In regard to (a) the Council accept the degree or diploma of any recognized University or Technical College in England or elsewhere, and examine from the Aeronautical paper of the Royal Society of Engineers, or from the Royal Society of Aeronautics of such University or Technical College.

In regard to (b) the Council does not insist upon a thesis necessarily being a record of original work but requires it to be of such a nature as to give an adequate idea of the Candidate's knowledge and experience in aeronautics or as allied subjects related to aeronautics.

In regard to (c) the Council has accepted photographs of rigid models, and airplane test and experimental plots whose work has been, or is likely to be, of value to the progress of technical development. It also has accepted long-standing workers in any branch of aeronautics whose experience satisfies them as to be considered as "Persons". In considering applications for Associate Fellowship it is not possible to lay down hard and fast rule but each case must be considered individually on its merits.

Suggests Air Mail Postcards

"Why not have a special rate for postcards by Air Mail?" writes Alfred Kirkwood from San Francisco. "The postcard rate is 1 cent. If we could get a rate of 25 cents for airmail it would cover the expense by Air Mail. That is at the rate of \$2.50 per airmail, \$25.00 per postcard, \$75,000 per day."

"Postcards are the one class of mail on which the Air Mail service can make large profits at reduced rates. But this traffic will not develop until the rates are reduced. They should be cut to 25 cent rate. This much correspondence that goes by air mail is now sent by surface mail," says A. J. McCall.

"The Air Mail world would soon be carrying a billion postcards a year at an average rate of 4 cents (five times), giving a revenue of \$60,000,000 a year from postcards alone. The Service would expand rapidly, and give us the strength in the air which we want."

Mr. McCall adds: "The Air Mail would soon be carrying a billion postcards per year, as Mr. Kirkwood hopefully suggests, the idea of charging a special rate for postcards carried by Air Mail appears to be quite sound, even if the present Air Mail letter postage were only reduced 50 per cent. The suggestion deserves serious consideration on the part of the Post Office Department."

AIRPORTS AND AIRWAYS

Long Island News

The firms of interest for the flying fraternity of Long Island are the two new shops which are being put out for experimental service with the Air Mail. The Air Mail in no way affects the flying of the members of the Legion, but it is known that if they could find a place where better than the D.H.s that they might buy a few. The Post Office Department has given out all definite specifications, except that they want a cruising speed of over 120 m.p.h., good maneuverability and clean lines. The two shops which are now competing at Carteret Field are quite different. One general article has just come by the members of many of the gadgets, accessories and appliances both manufacturers have been inspired by people who had pretty definite ideas as to what they wanted. The planes are being given high and low speed tests, cooling tests with full load (3000 lb.) and a nose authority run of 1000 ft. in 10 sec. The high speed and altitude record is 12,000 ft. in 1 min. These tests show that the manufacturers are the best. The Air Mail drivers will be sent with several other experimental planes to Chicago where they will be put on a trial run under service conditions. The observers for these preliminary tests are to be Fred Edward P. Wagner, Capt. J. E. Whetlock, Superintendent of the Eastern Division of the Legion, and C. Hanmer, Representative of the rotary club at Maywood.

The Carteret shop is called the Carter Fugue and is being flown by the head of the Carter Field staff, the popular Casey Jones. The Aerial Service Corporation's plane named the "Aerial Mercury" is being tested by Harvey Munson of that place. Since Mr. Munson was not present during or during the first stop at Carteret, the test flight is to be conducted by present experience whether it will do what he figured it would do.

The general appearance of the Mercury is rather pleasing but it is difficult to get in 30 cu. ft. of small compartment without making the fuselage somewhat bulky. The Mercury has a very long tail and is flying one of a series of Bleriot E.V. 8. The pilot can easily see over the tail section, having to turn out. The Carter Fugue is commonly called the "Tying express car" though the "Tying express car" might suit it better. The fuselage is very high and narrow and angular in appearance, but the pilot can only get good visibility directly ahead, while front view is poor. The tail is high and straight so one can look over the lower front wing and not down at quite a good angle even directly over the motor. Both planes have been built to get aeroplane and ease of maintenance. The Carter product has all the control surface interchangeable, while the Aerial Service Corp. has interchangeable surfaces except for the rudder.

The Mercury has many nice little gadgets on the cockpit for the benefit of the pilot. The rudder pedals are adjustable for height and depth. There is a leg-room from the exhaust which measures but six to the cockpit when the shoulder is set, a trim safety belt around the shoulder saves the pilot from movement but after a few moments a great deal more movement is possible. The seat is a leather one and has three points of adjustment so that one need not sit back against the back or sit forward in case of a crash. The control stick is made of aluminum, so that it will not affect the compass. A button on the control stick which operates on the instrument lights without taking the hand off the control stick. The fuse box has a cover to prevent moisture entering of the fuses while in the storage. All these little details show great planning and a knowledge of what is wanted.

All the controls are easily operated by trip dogs let into the bottom of the floor and wings. The tail skid is entirely externally mounted so that it can be easily inspected or re-

placed. A very neat one man dolly has been built for the plane which seems to work well and should save time and labor.

Both the Carter Fugue and the Mercury look like real ships. Probably no five pilots in the Air Mail will have exactly the same opinions about the various ships offered, except that they want a cruising speed of over 120 m.p.h., good maneuverability and clean lines. The two ships which are now competing at Carteret Field are quite different.

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Airport Services During 1924

Although the development of air services in Germany has been retarded by the restrictions of the Treaty of Versailles and by later interpretations of that Treaty, the year 1924 witnessed a marked increase in the number of passengers carried. The German Post, Berlin, the Reichspost, Berlin, and the Reichspostamt, Berlin, were created in 1923. The following tables are based on statements of the two principal German air services name, the Deutsche Luftpost and the Deutscher Lloyd.

Deutsche Luftpost.

Number of Passengers	Distance traveled (miles)
July 1, 1923 to date described	5,750
Total	1,000
Aug. 1, 1924 to date described	1,700
Total	2,100

Deutscher Lloyd.

Number of Passengers	Distance traveled (miles)
July 1, 1923 to date described	2,400,000
Total	2,400,000
Aug. 1, 1924 to date described	3,000,000
Total	5,400,000

Deutsche Luftpost. The two ships which are in operation at present are the *Deutschland* and the *Deutschland II*. The *Deutschland* has a range of 2800 miles and the *Deutschland II* has a range of 2500 miles.

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Two highly successful experiments in regular flights. Eight flights made by the two German air services from Berlin to Stockholm, and the Aero-Lite from Berlin to Copenhagen. Both journeys have been uneventful, regular uneventful for 1923-1924 for the first time. The construction of all-metal planes designed solely for the transportation of passengers, mail, and baggage, is contemplated as an important step forward in the development of German aviation. Another important step forward in the improvement of German aeronautics was the arrangement with the Prussian Railway approved Oct. 16, 1924, whereby air passengers were permitted to forward their luggage by train simply upon presentation of their tickets. This arrangement holds good, if revised, only within the country.

Russian Airway Expansions

A great expansion of the Russian air transport companies is planned for the summer of this year. The Deutsche will open a new route from Berlin to St. Petersburg via Moscow, Leningrad, and Archangel. A representative of the same company is making preparations for a line from Leningrad to Irkutsk in the Black Sea Provinces. These expansions are in line with the project of a linking up of Moscow with Tiflis and Peking. The principal use of the lines at present is for propaganda purposes.

Italy to Australia Flight Planned

The Italian Air Department announces that a flight from Rome, Italy to Sydney, Australia, via Japan, and return will be undertaken during the month of April. The Italian Government has contracted with the Italian Aeroplane Co. of Italy, chief of the Italian Air Force, to make it. He will fly by a Savoia 16-type biplane with a 400 hp engine, which will have a 50 hr. cruising range. The start will probably take place early in May.

Commander Prado's itinerary is from Rome to Egypt via Crete, then across the Red Sea and through the Suez Canal to Aden, to Port Said, Bangkok, the Mekong, Nanking, French Indo-China, China, Japan and from there by way of the Pacific across to Australia. The length of the trip is approximately 22,000 km.

Commander Prado as one of Italy's foremost seaplane pilots and he possesses particular experience in long sea flights.

French to U. S. Air Mail

Orders have been issued by the French Government concerning French air services and the air mail service between New York and Paris. The American post office plans to be of great interest to international business houses as it means a saving of from two to three days in the delivery of a letter along the Pacific Coast.

Letters for the new service are required to be printed uniformly in "Tout fait France." Air Mail letters and should have additional French postage to the amount of one franc for every 20 grams or fraction thereof for the New York-Clairefontaine section of the service, and additional two francs for each of the Western sections, terminating at Cheyenne, Wyo., and at San Francisco.

Efforts toward the establishment of the through service from Paris have been made for some time by the Aviation Committee of the American Legation, Paris Post No. 1, which has worked with the Society of the Friends of the Foreign Relations Committee and the National Association of the American Legion. Committee, composed of R. C. Wood, Chairman, and Colonel Charles A. Weston, presented our to Colonel Paul Emond, Assistant Postmaster-General in charge of the Air Mail, the most recent in time such a committee would offer the service. It is proposed to have the service ready within a day or two after the shipment of goods has been made by employing the Air Mail, however, this discrepancy may be overcome as the revenue will arrive at about the same time as the goods.

Wichita Maiden Tested



Z. S. Lyman, president Wichita Chapter N.A.A., completing a test flight on the performance of his new plane.

The latest design of Lloyd Starships was tested on March 13 at Wertheim. This plane was built by the Travel Air Manufacturing Co. and is the first of a series to be placed on the market. The tests were observed by the Wichita Chapter of the N.A.A. and justified by them. The average of three speed trials in a measured course with pilot and one passenger was 90 m.p.h. The 100 m.p.h. speed was obtained with an overload as follows: 1119 ft. at 90 m.p.h. and climbed to 580 ft. at 102 m.p.h. The climb with pilot and one passenger was 3000 ft. in 3 min. 6 sec. As this plane is equipped with a 60 hp engine, these figures appear to be very satisfactory. A full description of the new design is being prepared for aviators and will appear in an early issue.

Huntington News

By A. McMillan

Two more men have been added to the personnel of the Shiek McMillan Aircraft Company. Both are experts in their respective fields and will help to increase the service that the company has always endeavored to give its customers.

E. G. Gause has leased a large tract of land on the Eastern side of Huntington and is erecting thereon a first class flying field where he expects to operate a number of passenger planes. The field will be known as the Queen City Air Park.

Jamison James, ex K.A.F. pilot, has accepted a position with a large company to dust cotton in Georgia this summer.

A Flying Family

The Mexican family of Clever Field, Santa Monica, Calif., claims to be the largest family of pilots as it includes the father and three sons. The father J. W. (Daddy) Morris is 62 years of age and qualified as a pilot after he was 40. His three sons, Harry, George and Ruth, are engaged in the operation and construction of airplanes at Clever Field. J. W. Morris was modestly complimented by General Patrick on his becoming a pilot and his sons.

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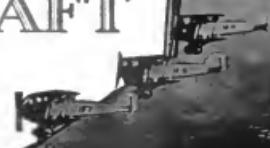
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